TECHNICAL MANUAL

UH-1H/V AND EH-1H/X AIRCRAFT PHASED MAINTENANCE CHECKLIST

CHANGE

NO. 24

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 16 April 2003

UH-1H/V and EH-1H/X AIRCRAFT PHASED MAINTENANCE CHECKLIST

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TM 55-1520-210-PM, 4 January 1983, is changed as follows:

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Remove pages Insert pages

A and B 2-17 and 2-18 A and B 2-17 and 2-18

2. Retain this sheet in front of manual for reference purposes.

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Joel B. Hulson

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CHANGE

NO. 23

HEADQUARTERS DEPARTMENT OF THE ARMY WASHINGTON, D.C., 8 November 2002

UH-1H/V and EH-1H/X AIRCRAFT PHASED MAINTENANCE CHECKLIST

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Remove pages Insert pages

A/(B blank) A and B 2-23 and 2-24 2-23 and 2-24

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TM 55-1520-210-PM C22

CHANGE

HEADQUARTERS DEPARTMENT OF THE ARMY WASHINGTON, D.C. 8 May 2002

NO. 22

TECHNICAL MANUAL

UH-1H/V and EH-1H/X AIRCRAFT PHASED MAINTENANCE CHECKLIST

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OZONE DEPLETING CHEMICAL INFORMATION

This document has been reviewed for the presence of Class I ozone depleting chemicals. As of Change 20, dated 4 January 1983, all references to Class I ozone depleting chemicals have been removed from this document by substitution with chemicals that do not cause atmospheric ozone depletion.

HAZARDOUS MATERIAL INFORMATION

This document has been reviewed for the presence of solvents containing hazardous materials as defined by the EPCRA 302 and 313 lists. As of Change 20, dated 4 January 1983, all references to solvents containing hazardous materials have been removed from this document with non-hazardous or less hazardous materials where possible.

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

1. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

Remove pages Insert pages A and B A and B 2-3 and 2-4 2-3 and 2-4

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CHANGE

NO. 21

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UH-1H/V and EH-1H/X AIRCRAFT PHASED MAINTENANCE CHECKLIST

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Remove pages	Insert pages
A/(B blank)	A and B
2–3 and 2–4	2-3 and 2-4

2. Retain this sheet in front of manual for reference purposes.

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TM 55-1520-210-PM C20

CHANGE

NO. 20

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 25 October 1999

TECHICAL MANUAL

UH-1H/V and EH-1H Aircraft PHASED MAINTENANCE CHECKLIST

DISTRIBUTION STATEMENT A Approved for public release; distribution is unlimited

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Remove pages	Insert pages
	A/(B blank)
1-3 and 1-4	1-3 and 1-4
2-31 and 2-32	2-31 and 2-32
2-35 and 2-36	2-35 and 2-36
2-59 and 2-60	2-59 and 2-60

2. Retain this sheet in front of manual for reference purposes.

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TM 55-1520-210-PM C19

CHANGE

NO. 19

HEADQUARTERS DEPARTMENT OF THE ARMY WASHINGTON, D.C., 13 January 1997

UH-1H/V and EH-1H/X Aircraft PHASED MAINTENANCE CHECKLIST

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TM 55-1520-210-PM, 4 January 1983, is changed as follows:

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Remove pages	Insert pages
2-21 and 2-22	2-21 and 2-22
2-37 and 2-38	2-37 and 2-38
2-61/(2-62 blank)	2-61/(2-62 blank)

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CHANGE

NO. 18

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UH-1H/V and EH-1H/X Aircraft PHASED MAINTENANCE CHECKLIST

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Remove pages	Insert pages
1-1 through 1-4	1-1 through 1-4
1-9 and 1-10	1-9 and 1-10
	1-10.1/(1-10.2 blank)
1-13 and 1-14	1-13 and 1-14
2-3 and 2-4	2-3 and 2-4
2-11 and 2-12	2-11 and 2-12
2-19/(2-20 blank)	2-19/(2-20 blank)
2-21 through 2-24	2-21 through 2-24
2-29 through 2-32	2-29 through 2-32
2-33/(2-34 blank)	2-33 and 2-34
2-35 through 2-38	2-35 through 2-38
2-49/(2-50 blank)	2-49/(2-50 blank)
2-51 and 2-52	2-51 and 2-52
2-59/(2-60 blank)	2-59 and 2-60
2-61/(2-62 blank)	2-61(262 blank)

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TM 55-1520-210-PM C17

CHANGE

NO. 17

HEADQUARTERS
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WASHINGTON, D.C., 15 NOVEMBER 1995

UH-1H/V and EH-1H Aircraft PHASED MAINTENANCE CHECKLIST

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Remove pages	insert pages
2-3 and 2-4	2-3 and 2-4
2-31 and 2-32	2-31 and 2-32
2-37 and 2-38	2-37 and 2-38

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WASHINGTON, D.C., 30 April 1992

UH-1H/V and EH-1H/X Aircraft PHASED MAINTENANCE CHECKLIST

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

1. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

Remove pages	Insert pages
1-13 and 1-14	1-13 and 1-14
2-3 and 2-4	2-3 and 2-4
2-9/2-10	2-9/2-10
2-27/2-28	2-27/2-28
2-29 and 2-30	2-29 and 2-30
2-31 and 2-32	2-31 and 2-32
2-51 and 2-52	2-51 and 2-52
2-53/2-54	2-53/2-54
2-61/2-62	2-61/2-62

2. Retain this sheet in front of manual for reference purposes.

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CHANGE NO. 15

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 22 February 1991

UH-1H/V and EH-1H/X Aircraft Phased Maintenance Checklist

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

l. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

Remove pages

Insert pages

1-1 through 1-4 2-21 and 2-22 1-1 through 1-4 2-21 and 2-22

2. Retain this sheet in front of manual for reference purposes.

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TM 55-1520-210-PM C 14

CHANGE)

HEADQUARTERS DEPARTMENT OF THE ARMY WASHINGTON, D.C., 6 September 1990

UH-1H/V and EH-1H/X Aircraft PHASED MAINTENANCE CHECKLIST

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

1. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

Remove pages	Insert pages
1-1 and 1-2	1-1 and 1-2
2-3 and 2-4	2-3 and $2-4$
2-15 through 2-32	2-15 through 2-32
2-35 through 2-42	2-35 through 2-42
2-46.1/2-46.2	2-46.1/2-46.2
2-47 through 2-52	2-47 through 2-52
2-54.1/2-54.2	2-54.1/2-54.2
2-57/2-58 through 2-61/2-62	2-57/2-58 through 2-61/2-62

2. Retain this sheet in front of manual for reference purposes.

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To be distributed in accordance with DA Form 12-31, PM requirements for UH-1H/V Helicopter, Utility, EH-1H Helicopter, Electronic Countermeasure, and EH-1X Helicopter, Electronic Countermeasure & Intercept.

TM 55-1520-210-PM C 13

CHANGE NO. 13

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 16 April 1990

UH-1H/V AND EH-1H/X AIRCRAFT PHASED MAINTENANCE CHECKLIST

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

l. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

Remove pages	Insert pages
2-23/2-24	2-23 and 2-24
2-27/2-28	2-27/2-28
2-31 and 2-32	2-31 and 2-32

2. Retain this sheet in front of manual for reference purposes.

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General, United States Army
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CHANGE NO. 12

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 22 November 1989

UH-1H/V AND EH-1H/X AIRCRAFT PHASED MAINTENANCE CHECKLIST

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

1. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

Remove pages	Insert pages
1-3 and 1-4	1-3 and 1-4
2-7/2-8	2-7/2-8
2-23/2-24	2-23/2-24
2-27/2-28	2-27/2-28
2-31 and 2-32	2-31 and 2-32
2-49/2-50	2-49/2-50
2-55 and 2-56	2-55 and 2-56

2. Retain this sheet in front of manual for reference purposes.

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TM 55-1520-210-PM C 11

CHANGE NO. 11

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 22 August 1988

UH-1H/V AND EH-1H/X AIRCRAFT PHASED MAINTENANCE CHECKLIST

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

1. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

Remove pages	Insert pages
2-17 and 2-18 2-27/2-28	2-17 and 2-18 2-27/2-28
2-31 and 2-32	2-31 and 2-32
2-46.1/2-46.2	2-46.1/2-46.2

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URGENT

CHANGE NO. 10

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 6 June 1988

UH-1H/V AND EH-1H/X AIRCRAFT PHASED MAINTENANCE CHECKLIST

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

l. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

Remove pages	Insert pages
1-9 and 1-10	1-9 and 1-10
2-3 and 2-4	2-3 and $2-4$
2-7/2-8	2-7/2-8
2-13/2-14	2-13/2-14
2-17 and 2-18	2-17 and 2-18
2-45/2-46	2-45/2-46

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TM 55-1520-210-PM C 9

CHANGE NO. 9

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 26 February 1986

UH-1H/V AND EH-1H/X AIRCRAFT PHASED MAINTENANCE CHECKLIST

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

1. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

Remove pages	Insert pages
1-3 and 1-4	1-3 and 1-4
1-9 and 1-10	1-9 and 1-10
1-13 and 1-14	1-13 and 1-14
2-3 and 2-4	2-3 and 2-4
2-46.1/2-46.2	2-46.1/2-46.2
2-47 and 2-48	2-47 and 2-48
2-51 and 2-52	2-51 and 2-52

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TM 55-1520-210-PM C 8

CHANGE NO. 8

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 15 October 1985

UH-1H/V AND EH-1H/X AIRCRAFT PHASED MAINTENANCE CHECKLIST

TM 55-1520-210-PM, 4 January 1983, is changed as follows:

1. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

Remove pages	Insert pages
1-15 and 1-16 2-35 and 2-36 2-41 and 2-42 2-53 and 2-54	1-15 and 1-16 2-35 and 2-36 2-41 and 2-42 2-53/2-54
2-59/2-60 2-61/2-62	2-54.1/2-54.2 2-59/2-60 2-61/2-62

2. Retain this sheet in front of manual for reference purposes.

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URGENT

LIST OF EFFECTIVE PAGES

Insert latest changed pages; dispose of superseded pages in accordance with regulations.

NOTE: On a changed page, the portion of the text affected by the latest change is indicated by a vertical line, or other change symbol, in the outer margin of the page. Changes to illustrations are indicated by miniature pointing hands. Changes to wiring diagrams are indicated by shaded areas.

Dates of issue for original and changed pages are:

Original 0 Change 1 Change 2 Change 3 Change 4 Change 5 Change 6 Change 7 Change 8 Change 9 Change 10 Change 11	04 January 1983 31 January 1983 18 March 1983 18 April 1983 14 March 1984 18 April 1985 19 April 1985 12 July 1985 15 October 1985 26 February 1986 06 June 1988 22 August 1988	Change	12 22 November 1989 13 16 April 1990 14 06 September 1990 15 22 February 1991 16 30 April 1992 17 15 November 1995 18 01 July 1996 19 13 January 1997 20 25 October 1999 21 2 October 2000 22 8 May 2002 23 8 November 2002 24 16 April 2003
Page	*Change	Page	*Change
No.	No.	No.	No.
Cover		2-6 (blank) 2-7	

^{*}Zero in this column indicates an original page.

Page	*Change	Page	*Change
No.	No.	No.	No.
2-29	14 18 20 18 8 20 14 19 0 0 14 5 0 10 10	2-47 and 2-48	

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 4 January 1983

UH-1H/V AND EH-1H/X AIRCRAFT PHASED MAINTENANCE CHECKLIST

WARNING

CERTAIN INSPECTIONS ARE MANDATORY SAFETY-OF-FLIGHT REQUIREMENTS, AND THE INSPECTION INTERVALS CANNOT BE EXCEEDED. IN THE EVENT THESE INSPECTIONS CANNOT BE ACCOMPLISHED AT THE SPECIFIED INTERVAL, THE AIRCRAFT CONDITION STATUS SYMBOL WILL BE IMMEDIATELY CHANGED TO A RED X. MANDATORY SAFETY-OF-FLIGHT INSPECTION ITEMS ARE PRINTED IN BOLD FACE TYPE.

NOTE

INSPECTION ITEMS CONTAINED IN THIS MANUAL ARE CONSIDERED THE MINIMUM REQUIREMENTS FOR PERFORMING PHASED MAINTENANCE AND MUST BE PERFORMED. THE CUMULATIVE EFFECTS OF INSPECTION DEFERRALS ARE UNKNOWN AND COULD RESULT IN CATASTROPHIC FAILURE OR INCREASED MAINTENANCE AT A LATER DATE. THEREFORE, THE USE OF SPECIAL LETTERING TO EMPHASIZE MANDATORY SAFETY-OF-FLIGHT INSPECTION ITEMS IS NOT TO BE CONSTRUED AS AUTHORITY FOR DEFERRAL OF OTHER INSPECTIONS.

^{*} This manual supersedes TM 55-1520-210-PM, 20 July 1978, including all changes.

SECTION I - GENERAL INFORMATION

- **1-1. PHASED SCHEDULE.** This phased maintenance inspection checklist contains requirements for inspection of the UH-1H/V and EH-1H/X aircraft on phased schedule having a 900 hour (flight hours) cycle with 150 hour phases. Each requirement included herein is designated for accomplishment at least once, but not more than six times during the 900 hour cycle.
- **EXCEEDING THE PHASED SCHEDULE.** The phased maintenance 1-2. inspection intervals designated are the maximum and shall not be exceeded except in actual operational emergencies as explained herein. It is the Commander's responsibility to determine (on an individual aircraft basis) when inspection intervals may be exceeded. For this purpose, operational emergencies are conditions of combat or conditions of disaster which necessitate flight to evacuate aircraft or personnel. Those inspections annotated by a C in the Inspect Phase No's column along with the DA Form 2408-18 (Equipment Inspection Record) items that are due are considered the MINIMUM mandatory Combat maintenance inspection requirements for helicopters scheduled for imminent deployment to or stationed in a combat environment. Under no circumstances will two Combat Phase inspection be performed sequentially. When inspections are delayed to meet emergency requirements, Commanders will assure that the I aircraft status symbol reverts to a red "X" and that delayed inspections are accomplished immediately upon termination of the actual emergency. When unusual local conditions (utilization, type of mission personnel, periods of inactivity, environmental conditions, etc.) dictate, it is the prerogative and responsibility of the Maintenance Officer to increase the scope and/or frequency I of maintenance as necessary to insure safe operation (TM 1-1500-328-23).
- **1-3. MAINTENANCE ACTIVITIES**. The inspections prescribed by this checklist will be accomplished at specified phases by Aviation Unit Maintenance (AVUM) activities with assistance of Aviation Intermediate Maintenance (AVIM) and Depot Maintenance activities when required.
- **1-4. LIMITATIONS**. The checklist does not contain instructions for repair, adjustment or other means of rectifying conditions. Neither does it contain special tolerances, limits or instructions for special troubleshooting to find causes for malfunctions. Such data will be obtained from the latest issue of the aircraft (TM 55-1520-210-23) series maintenance manuals.
- **1-5. CHANGEOVER TO THE PHASED MAINTENANCE SYSTEM.** Changeover shall be accomplished in accordance with instructions provided in TB 55-1500-337-24 entitled, "Phased Maintenance System for Army Aircraft". The requirements of this TB must be accomplished prior to implementation of Phase 1 inspection requirements specified in this checklist.
- **1-6. PRE-INSPECTION MAINTENANCE TEST FLIGHT (MTF).** A preinspection MTF to duplicate nonhazardous equipment problems, determine unsatisfactory conditions, determine equipment operations problems, etc., is recommended prior to start of aircraft disassembly for phased maintenance inspection. The decision to perform the pre-inspection MTF, however, shall be the responsibility of the unit Maintenance Officer.

- 1-7. SPECIAL INSPECTION, CALENDAR INSPECTION AND LUBRICATION REQUIREMENTS. Special inspection, calendar inspection and lubrication requirements contained in C(M 55-1520-210-23) and those listed on the aircraft DA Form 2408-18 shall be reviewed and accomplished in accordance with the "inspection due" requirements specified in those documents.
- 1-8. TIME BETWEEN OVERHAUL (TBO) AND RETIREMENT LIFE ITEMS CHECK. Prior to start of the applicable phased maintenance inspection, a check will be made of components and their remaining operating hours prior to removal. The latest issue of the aircraft, TM 55-1520-210-23 and DA Form 2408-16, shall be referred to for a complete listing of components and their TBO and retirement life.
- **1-9. USING THE PHASED INSPECTION CHECKLIST**. For use of the phased inspection checklist, refer to DA Pam 738-751.
- 1-10. FINAL RECORDS CHECK. After all corrective actions have been completed and following completion of the phased inspection, the technical inspector or designated supervisor shall verify that all applicable forms and records have been properly updated. Any fault not corrected will be carried forward to a new DA Form 2408-13 or reentered on DA Form 2408-14. A final records checklist (Table 1-2) is provided to ensure forms and records have been inspected for completeness and accuracy prior to release of the aircraft from the phased maintenance inspection. The inspector verifying the final records check shall enter his initials adjacent to the indicated form or record on the Final Records Checklist. The initials entered shall be registered on the Signature Sheet (Table 1-1) adjacent to that person's signature.

- **1-11. SIGNATURE SHEET.** All personnel performing inspection and/or maintenance tasks shall place their signatures and initials on the signature sheet (Table 1-1). The purpose of the signature sheet is to provide a correlation between initials entered on the individual checklist sheets and the actual names of the personnel accomplishing these tasks.
- **1-12. MAINTENANCE OPERATIONAL CHECKS.** After the completion of any required corrective actions to any of the components of the function system of the aircraft, maintenance operational checks (MOC) shall be performed on that system to determine the effectiveness of maintenance actions performed and to verify to proper operation of that system. These MOC shall be performed in accordance with TM 1-1500-328-23. Copies of DA Form 240813-1 (Figure 1-1) and DA Form 2408-13-2 (Figure 1-1.1) may be used to record and sign off the MOC performed.
- **1-13. MAINTENANCE TEST FLIGHT**. When all required inspections in Section II have been accomplished and initialed in accordance with the above procedures, a daily inspection in accordance with the TM specified in Section II will be preformed on the aircraft to permit a maintenance test flight (MTF) to be made. The MTF shall be performed in accordance with the requirements of TM 55-1520-242-MTF and TM 1-1500-328-23, using the MTF Form in the MTF Technical Manuals. A suggested maintenance test flight checksheet (Figure 1-5) and Rotor Smoothing Record (Figure 1-6) are provided at the end of Section I.
- **1-14. CHECKLIST DISPOSITION.** The completion of each phased maintenance shall be recorded on DA Form 2408-13 and DA Form 2408-15 as prescribed by DA PAM 738-751. The signed checklist, together with all continuation sheets shall be attached to DA Form 2408-13, and filed for the six month period as required by DA PAM 738-751.

- **1-15. INSPECTION AREAS.** Figures 1-2 and 1-3 show the inspection areas of the UH-1H/V and EH-1H/X aircraft. These areas are titled as shown below. Figure 1-4 shows the location of access doors and panels which require removal at various phased maintenance inspections. Access panels and doors are identified by number of the text where applicable. Additional panels and doors may be removed as required to facilitate inspection requirements.
- **1-15.1** This TM checklist may contain inspection requirements applicable to specific equipment not installed on your aircraft. Those requirements should be disregarded

- **1-16**. Deleted.
- 1-17. REPORTING ERRORS AND RECOMMENDING IMPROVEMENTS. You can help improve this manual. If you find any mistakes or if you know of a way to improve the procedures, please let us know. Mail your letter, DA Form 2028 (Recommended Changes to Publications and Blank Forms), or DA Form 2028-2 located in the back of this manual direct to: Commander, U.S. Army Aviation and Missile Command (AMCOM), ATTN: AMSAMMIC-LS-LP, Redstone Arsenal, AL 35898-5230. You may also submit your recommended changes by E-mail directly to Is-Ip@redstone.army.mil, or by fax at 256-842-6546 or DSN 788-6546. A reply will be furnished directly to you. Instructions for sending an electronic 2028 may be found at the back of this manual immediately preceding the hardcopy DA Forms 2028.

AREA NO.	AREA TITLE
1	Aircraft Exterior
2	Nose Area
3	Forward Radio/Battery Compartment
4	Cockpit Interior
5	Cabin Interior
6	Under Floor Cockpit/Cabin
7	Lower Pylon Area (Via Cabin Interior)
8	Upper Pylon Area (Via Cabin Roof)
9	Main Rotor and Mast Area
10	Engine Air Induction Area
11	Engine Compartment
12	Cabin Roof
13	Cabin Sides, Bottom and Landing Gear
14	Under Cabin Pylon Area (Hell Hole)
15	Mid Fuselage Under Engine Deck
16	Electronic Comm. Compartments
17	Engine Area Exterior
18	Tailboom Interior
19	Tail Rotor Drive Train Area
20	Tail Rotor and Gearbox Area
21	Oil Cooler Aft Battery Compartment
22	Heater Compartment

Table 1-1. Signature Sheet (Sheet 1 of 3)

Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Maintenance Supervisor	Initial
Signature of Technical Inspector	Initial
Signature of Maintenance Officer	Initial

Table 1-1. Signature Sheet (Sheet 2 of 3)

Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Maintenance Supervisor	Initial
Signature of Technical Inspector	Initial
Signature of Maintenance Officer	Initial

Table 1-1. Signature Sheet (Sheet 3 of 3)

Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Maintenance Supervisor	Initial
Signature of Technical Inspector	Initial Initial
Signature of Maintenance Officer	

Table 1-2. Final Records Checklist

This checklist is provided to insure the indicated forms and records have been inspected for presence, completeness, legibility, and accuracy prior to releasing the aircraft from a phase inspection. Vertification of inspection will be indicated by placing the initials of the inspector in the appropriate initial block (refer to DA PAM 738-751).

AIRCRAFT LOG BOOK	INITIAL	HISTORICAL RECORDS	INITIAL
DA FORM 2408		DA FORM 2408-5	
DA FORM 2408-12		Delete	
DA FORM 2408-13		DA FORM 2408-15	
DA FORM 2408-14		DA FORM 2408-16	
DA FORM 2408-18		DA FORM 2408-17	
TM 55-1520-210-PMD		DA FORM 2408-19	
TM 55-1520-242-MTF		DA FORM 2408-20	
LOCALLY REQUIRED FORMS		LOCALLY REQUIRED FORMS	

PRODUCTION CONTROL	15.1171.51	CHALITY CONTROL	INITIAL
RECORDS	INITIAL	QUALITY CONTROL	HALLIAL
FLOW CHART		TBO FILE	
STATUS BOARD		QA FILE	
WORK ORDER FILE		SERIAL NUMBER FILE	
MWO FILE		AOAP FILE	
Delete		INVENTORY RECORDS	
1352 REPORTS		WEIGHT AND BALANCE	
LOCAL RECORDS		MSG FILE	
<u> </u>		DA FORM 2410 SUBMITTED	
		LOCAL RECORDS	

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DA FORM 2408-13-1, OCT 91

AIRCRAFT INSPECTION AND MAINTENANCE RECORD For use of this form, see DA PAM 736-751; the proponent agency is DCSLOG

Figure 1-1. DA Form 2408-13-1

	C	DATE		'				
		2. SERIAL NUMBER		3. SYSTEM CODE	<u> </u>	4. TIME		
1. STAT	rus	5. FAULT DATE		6, FAULT NUMBER		7		
7. FAU	LT							
8.STA	9. FAULT			IO. ACTION	11.	. PIO	12. MMH	
	<u> </u>							
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DA FORM 2408-13-2, NOV 91

RELATED MAINTENANCE ACTIONS RECORD

For use of this form, see DA PAM 738-751; the proponent agency is DCSLOG

Figure 1-1.1. DA Form 2408-13-2

1-10.1/(1-10.2 blank) C18

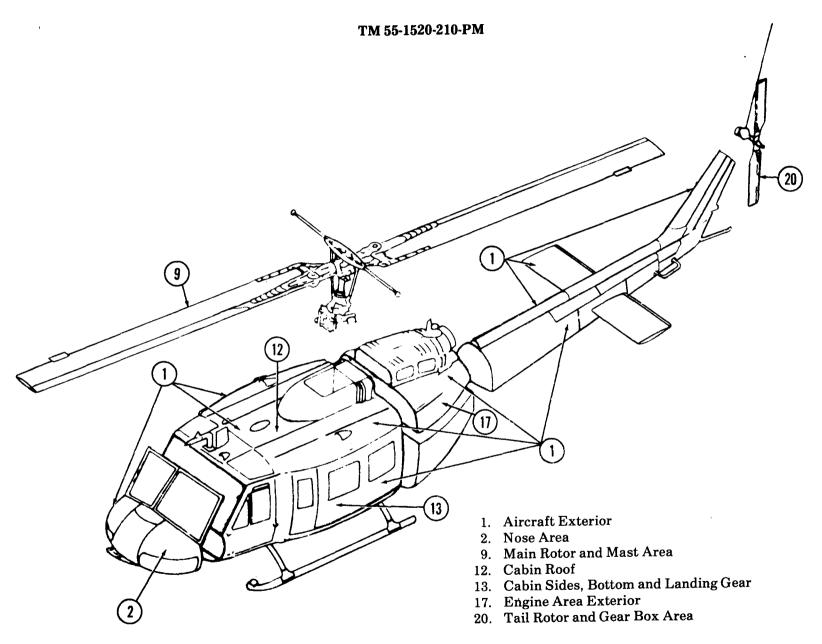


Figure 1-2. Exterior Inspection Areas

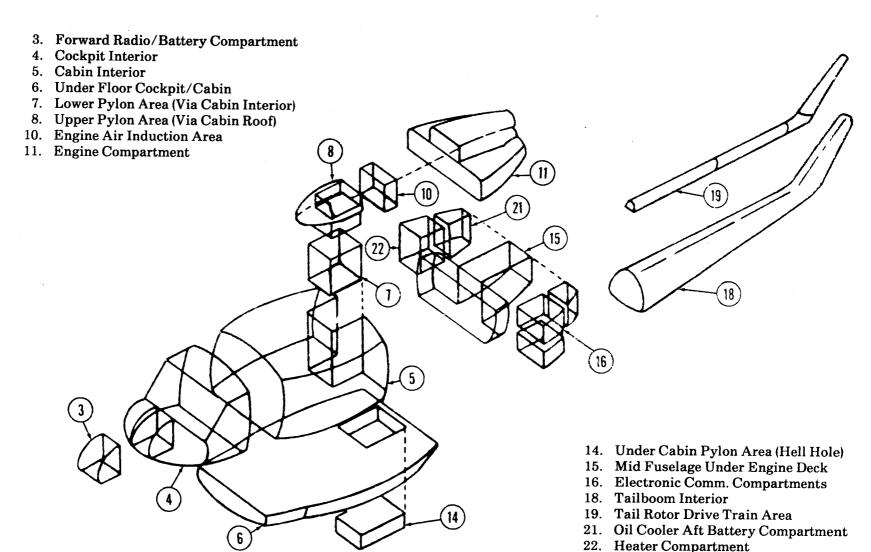


Figure 1-3. Interior Inspection Areas

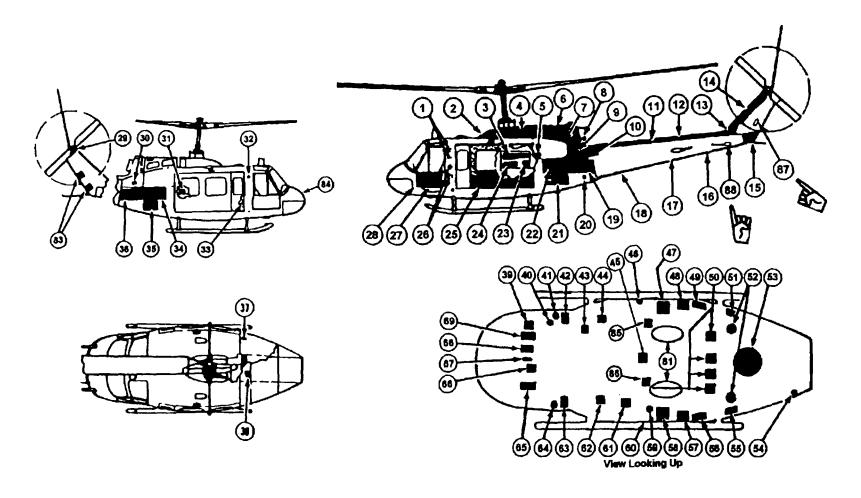


Figure 1-4. Model UH-1H/V and EH-1H/X Access and Inspection Provisions (Sheet 1 of 2)

1-13 C18

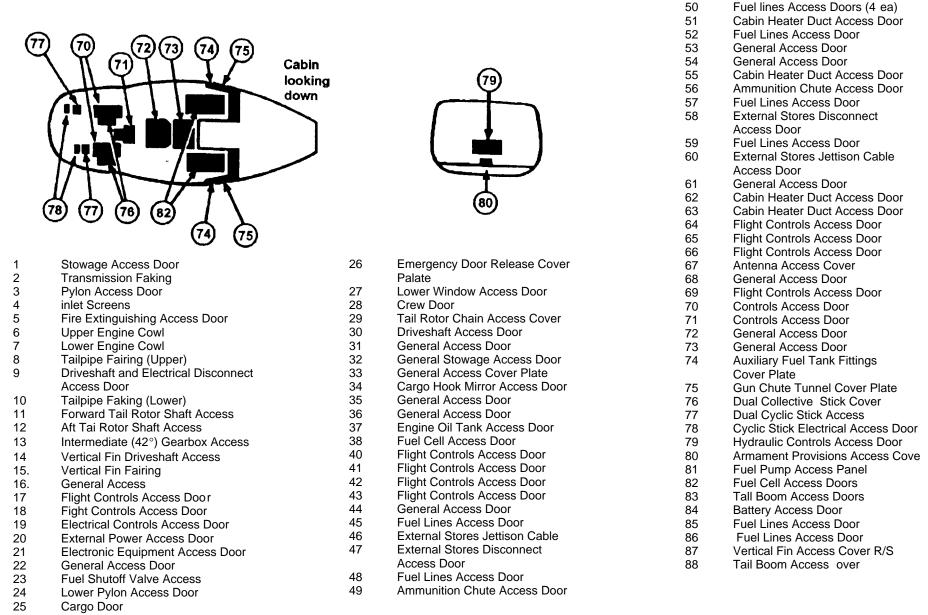


Figure 14. Model UH-1H/V and UH-1H/X Access and Inspection Provisions (Sheet 2 of 2)

C 8

1-15

UH−1 CE	H.M.V. & EH-1H/x
PURPOSE OF MIT FAT	ACFT MODEL & SN DATE
PILOT	UNIT
F (CO)	
SYMBOLS V= SATISFACTORY X = DEFI	CIENCY
Prior to Test Flight	2. Power Check
1. Forms and Records	3. Control Responses Checks
2. Flight Readiness Insp.	4. Pylon Mounts Checks
Weight and Balance	5. Engine Response
4. Engine Baseline Data	6. Power Cylinder
TO N1 EGT	7. Low RPM Hover
Starting Engine	8. Hover in Emergency 9. Torque MeterPSI
1. Press to Test Lights	9. Torque Meter PSI LEVEL OFF CHECKS
 Fire Warning Light Caution Panel Lights 	1. Eng Oil PressTemp
4. Throttle System Cushion	2. XMSN Oil Press Temp
OPEN CLOSED	3. EGT°C
ENGINE RUNUP	4. Airspeed Indicators
l. Engine Idle % Nl	INFLIGHT CHECK
2. Emerg Gov Switch	1. Control Rigging
3. Hydraulic System	2. Autorotation RPM
4. Fuel Boost Pumps	3. Hydraulics Off
Right PSI Left PSI	4. TEAC: PA
5. Bleed Band Operation	FAT TQ
OPEN CLOSE %N1FAT	N1 EGT
6. Variable Inlet Gui de Vanes	
BEGIN TO OPEN % N1FAT	6. Vibration Analysis
7. Fuel Quantity Gage	7. Cyclic Rigging
8. Pitot Heater	8. Fuel Consumption Initiate TIME FUEL
9. Spare Inverter AB AC BC	9. Instruments
10. Main GEN VDC	Altimeters Att. Ind.
11. STBY GENVDC	VSI Stby Comp
12. Main Inverter	RMI Clock
AB AC BC	Turn & Slip
13. Bleed Air Heater	10. Comm/NAV Radios
14. Deice Operation	UHF VHF FM1 FM2
15. Low RPM Warning	VOR ADF LOC/GS
Off On	MB XPONDER
16. GOV INCR/DECR	MODE C
Full Incr DECR	11. Fuel Consump. Complete
Travel Time SEC 17. High RPM Warning	Time Fuel AFTER LANDING/ENGINE SHUTDOWN
OFF ON	1. EGT
18. Force Trim System	2. Eng Oil Press Temp
19. Collective Friction	3. XMSN Oil Press Temp
UP DOWN	4. Eng Idle %N1
20. Eng Oil PressTemp	5. Battery
21. XMSN Oil PressTemp	6. Eng Oil Press Lite
22. Fuel PressPSI	7. XMSN Oil Press Lite
23. Torque Press PSI	8. N1 Coastdown Time Sec
24. EGT °C	9. Emerg. Collective Accum.
25. Altimeters P CP BEFORE TAKEOFF	10. Post Flight 11. Forms & Records Complete
1. HIT Check	SPECIAL REQUIREMENT (LIST)
HOVER CHECK	1.
1. Takeoff to hover	2.

	ROTOR SMOOTHING RECORD								
RED BLADE SERIAL NUMBER						TE BL/	ADE NUMBE	R	
ADJUSTMENT NUMBER	TAB	ROLL	BALANCE	EFFECT	ADJUSTMENT NUMBER	TAB	ROLL	BALANCE	EFFECT
1					1				
2					2				
3					3				
4					4				
5					5			. =	
				REM.	ARKS				

Figure 1-6. Suggested Format of Rotor Smoothing Record

PILOIS SIGNATURE

C 8 1.16

SECTION II - INSPECTION CHECKLIST

NOTE

PRIOR TO START OF THE PHASED MAINTENANCE INSPECTION, IT IS RECOMMENDED THAT A PRE-INSPECTION MAINTENANCE TEST FLIGHT (MTF) BE CONDUCTED. ACCOMPLISHMENT OF THE MTF SHALL BE DETERMINED BY THE UNIT MAINTENANCE OFFICER. THE PRE-INSPECTION MTF SHOULD BE CONDUCTED BY A MAINTENANCE TEST PILOT FOLLOWING A REVIEW OF THE AIRCRAFT FORMS AND RECORDS AND A BRIEFING FROM THE REGULAR FLIGHT CREW OF THE AIRCRAFT. THE MTF IS RECOMMENDED TO ASSESS THE AIRCRAFT PERFORMANCE AND IDENTIFY DEFICIENCIES THAT SHOULD BE CORRECTED WHILE THE AIRCRAFT IS UNDERGOING PHASED INSPECTION.

			Area Name and No.	Aircraft Serial No.	Date
PHASE N	0				
Inspect Phase No's	Inspection Requirement	Status	Faults and/or Remarks	Action Taken	Initial
ALL	Prior to inspection, check aircraft forms and records for deficiencies (use Table 1-2 for reference to aircraft forms and records).				
ALL	2. Clean engine in accordance with TM 55-2840-229-23				
ALL	Clean aircraft in accordance with latest issue of the aircraft AVUM AVIM maintenance manuals.				
	4. Deleted				
ALL	4 1 Aircraft with ODDS check all electrical chip detectors for metal accumulation, clean, perform functional check and reinstall.				
ALL	4.2 Inspect the three mounting grooves of chip detector probe for wear at the detent of the 90 and 42 degree gear box chip detectors and the three pins located in the chip detector valve for looseness.				
ALL	4.3. Inspect ODDS chip detector (Blade Type) for wear on blade or chip detector valve.				
ALL	5. Aircraft without ODDS, check all electrical chip detectors (except engine) for metal accumulation, clean, perform functional check and reinstall. Refer to TM 55-2840-229-23 for engine chip detector check.				

"FOD REMINDER"

Check work area for tools and parts after completion on maintenance and inspection.

Phase	No		GENERAL (CONT)	lame and No.		Aircraft Serial No.	Date		
Inspect Phase No.		Inspection R	lequirements	Status	Faults and/or l	Remarks	Action Taken		Initial
3,6	6.	Defuel aircraft in accordance with TM 55-1520-210-23 prior to removal of floor panels.							
1,2, 4,5	7.	of phased inspections	ully serviced prior to start s. If maintenance is to be required defueling, this ferred until after such leted.						
ALL C	8.	electrical equipment avionics publications	spections, check and test as required in applicable s. Any faults discovered is shall be entered on DA 8-13-1-E.						
ALL	9.	and test as required	system inspection checks in applicable armament ults discovered during the entered on DA Form 2408-			11-1			
ALL C	10.	Perform engine ex- functional test in ac manuals. Any fault inspections shall be e 13-1/2408-13-1-E.	xhaust gas temperature cordance with applicable s discovered during the entered on DA Form 2408-						

"FOD REMINDER"
Check work area for tools and part after completion of maintenance and inspection.

i	ASE NO	Area Name and No. AIRCRAFT EXTERIOR - 1			Aircraft Serial No. D			
Inspect Phase No's	Inspection (Requirements	Status	Faults and/or R	lemarks	emarks Action Taken		leifini
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PH	PHASE NO NOSE AREA - 2		ame ar	ame and No.		rcraft Serial No.	Date	2
Inspect Phase No:s	Inspection I	Requirements	Status	Faults and/or	r Remarks	Action Taken		Inilial
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			<u> </u>					ļ
			 -	<u></u>				
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2,4,6	2. Wiper blade arms	for condition, security	}					l
	and proper adjustment.							
3 & 6	Remove windshield wip	per motor and converter						
	assy, disassemble, clean	, inspect and service con-	<u> </u>					ļ
	verter.		 					├ ──
			 					
								
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	Area h		ame and No.		ircraft Serial No.	Date	
Phase	No	FORWARD RADIO/BAT	TERY	COMPARTMENT - 3			
Inspect Phase No.	Inspection I	Requirements	Status	Faults and/or R	emarks	Action Taken	Initial
ALL	1. Electrical wiring in nose compartment and behind instrument panel for chafing, deterioration of insulation and connector seals, and security of connections.						
ALL	2. Electrical equipment shock mounts for deterioration, free throw, bottoming and security. Grounding straps or bands for damage and security of connections.						
ALL	3. Heat/defog ducts and security.	and valve for damage					
ALL	4. Lines and hoses be panel for loose conn						

"FOD REMINDER"
Check work area for tools and parts after completion of maintenance and inspection.

PH	PHASE NO		Area Name and No. COCKPIT INTERIOR - 4			Aircraft Serial No. Da		Date	5
Inspect Phase No's	pect			Status	Faults and/or F	Remarks	Action Taken		Inilial
ALL	1.	Crew door jettison mechanism functionally check with doors closed. Hinge pins for wear, corrosion and distortion.							
ALL	2.	Release cables security and adeq	for chafing, damage, uate lubrication.						
ALL	3.	Door jettison han copper safety wire	dles properly wired with e.						
ALL	4.	Seat adjustment positive movement tion.	mechanisms for wear, nt, locking and lubrica-						
3,6	5.	Inspect seats movement.	for positive recline						

Phase	No		Name an OR 4			Aircraft Serial No.	Dat	te
Inspect Phase No.		Inspection Requirements	Status	Faults and/or	Remarks	Action Taken		Initial
3,6			d or					
		access). (Access panels 70, 71, and 76, Fig 14).	-					
ALL	7.	Electrical wiring for chafing, deterioration an security (pedestal console).	d					
		security (pedestal console).					4	
			<u> </u>					
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ALL	8.	Check circuit breakers, switches and knobs for security and proper operation.)r					<u> </u>
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"FOD REMINDER"

Check work area for tools and part after completion of maintenance and inspection.

PHASE NO CABIN INTERIOR - 5 Inspect Phase No's Inspection Requirements 3,6 1. Cabin floor panels for cracks, dents, delamination and security. Check that aircraft has been defueled (TM 55-1520-210-23-1) prior to removal of floor panels over the fuel cells and inspect for corrosion. (All floor Access Panels, Fig 1-4.) 3,6 2. Cabin structure for damage, cracks and corrosion (soundproofing removed and plates, panels, and doors opened for access). (All Access Panels, Fig 1-4).		ame ar	ame and No. Airc		rcraft Serial No.	Date	2		
Inspect Phase No's		Inspection F	Requirements	Status	Faults and/or	Remarks	Action Taken		Initial
3,6	1.	delamination and craft has been de 23-1) prior to remothe fuel cells and it	security. Check that air- fueled (TM 55-1520-210- oval of floor panels over aspect for corrosion. (All						
3,6	2.	corrosion (sound) plates, panels, an	proofing removed and doors opened for ac-						
3,6	3.	Cargo door windo functionally check	ow jettison mechanisms ed.						
ALL	4.	Inspect cargo door worn or bent.	r retainers for cracks,						

PH	ASE NO	UNDER FLOOR OF CO	Name at		A	ircraft Serial No.	Date
Inspect Phase No's	Inspec	tion Requirements	Status	Faults and/or I	Remarks	Action Taken	Initial
ALL	corrosion (flo	cture for damage, cracks and or panels removed for access). s 72 and 73, Fig 1-4.)					
ALL	accumulation.	oor for evidence of moisture Drain holes for clogged con- s Panels 72 and 73, Fig 1-4.)					
3,6 C	3. Collective fric Panel 76, Fig 1-	tion liners for wear. (Access					
ALL C	tubes, links, arms, jacksha corrosion, dai	linkages, including pushpull bellcranks, idlers, levers, fts, force gradients, etc., for nage and security. (Access 72 and 73, Fig 1-4.)					
ALL C	control linka	hings and rod end in flight ges for excessive play and ess Panels 70, 71, 72 and 73,					

ŧ		NO	Area N UNDER FLOOR OF CO			A	Aircraft Serial No.		•
Inspect Phase No's		Inspection I	Regulrements	Status	Faults and/or F	Remarks	Action Taken		Initiat
ATT	C	7	61						
ALL C	6.		kage for damage, wear, ation. (Access Panels 70,						
	i	71, 72 and 76, Fig 1-4	£.)						
<u> </u>									
}	j					*			
ALL	7.	Electrical wiring for	or chafing, deterioration						
! .			ess Panels 71, 72 and 73,						
	İ	Fig 1-4.)							
}						· · · · · · · · · · · · · · · · · · ·			
3,6		••				*			
3,6	8.	Panel 72, Fig 1-4.)	urity and damage. (Access						
		1 unor (2, 1 ig 1-4.)							
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PH	ASE	NO	Area N LOWER PYLON AREA			^	ircraft Serial No.	Date	
Inspect Phase No s		Inspection I	Regulrements	Status	Faults and/or F	l	Action Taken		Initial
ALL	1.	Transmission movand deterioration. 24, Fig 1-4).	int boots for cuts, tears (Access Panels 3, 23 and						
2,4,6 C	2.	inspect re each) for deterioration ty. (Access Panels 3,	esilient pylon mounts (5 on, cleanliness and securi- 23 and 24, Fig 1-4.)						
ALL	3.	Friction dampers (security. (Access Par	2 each) for damage and nels, 23 and 24, Fig 1-4.)						
ALL	4.	and fifth mount sup	tural supports (4 places) port fitting (1 each) visual- rrosion. (Access Panels 23						
ALL	5.	Lift link for corros ty. (Access Panel 2-	sion, damage and securi- 4, Fig 1-4.)						

PHASE NO	0	LOV	Area Name and No. VER PLYON AREA (VIA N INTERIOR) 7 (CONT)	Aircraft Serial No.	Date
Inspect Phase No's	Inspection Requirements	Status	Faults and/or Remarks	Action Taken	Initial
2,4,6 C	6. Inspect Lift Link Bearings. Inspect transmission clevis lugs (both right and left through access panels 24, Fig 1–4) for wear and cracks on the outboard side of the left transmis sion clevis lug. Check for proper gap between washer on left link attaching bolt and bushing sleeve.				
ALL C	7. Lift beam visually for cracks. (Access Panels 24 and 79, Fig 1-4 and Hell Hole.)				
ALL C	8. Power turbine governor controls for damage, wear and security. (Access Panels 23 and 24, Fig 1–4.)				
3,6 C	9. Hydraulic filter element (either paper or metal) replaced. (Access Panel 79 Fig 1–4.)	,			
ALL	Electrical wiring for chafing, deterioration of insulation and connector seals and security of connections (Hell Hole).				

"FOD REMINDER"

Check work area for tools and parts after completion of maintenance and inspection.

Phase	No	Area M LOWER PYLON AR INTERIOR) — 7 (CC				Aircraft Serial No.	Dat	e
Inspect Phase No.	Inspection R	equirements	Status	Faults and/or	Remarks	Action Taken		Initial
6	11. Remove and replace pin. (Access panel 24, Fig	droop compensator shear						
	pin. (Access paner 24, Fig	g 1 -4) .						
					<u></u>			
ALL C	12. All fuel supply lines	for chafing, damage and			~~~			
C	12. All fuel supply lines leaks. Self-sealing li swelling, blistered a soaked with fuel or	areas that appear to be						
	plies). (Access panel	23, Fig 1-4).						
							· · · · · · · · · · · · · · · · · · ·	
								
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"FOD REMINDER"

Check work area for tools and part after completion of maintenance and inspection.

Phase	e No UPPER PY	Area Name and No. LON (VIA CABIN R	OOF) — 8	Aircraft Serial No.	Date
Inspect Phase No.	inspection Requirements	Status	Faults and/or Remarks	Action Taken	Initial
ALL	Hydraulic reservoir, filler cap and condition. (Access panel 2, Fig 1-4).	strainer for			
ALL C	2. Generator/alternator drive quill mag removed and visually checked for cor Check vent on generator quill case for (Access panels 2 and 3, Fig 1-4).	ntaminants.			
ALL	3. Generator/alternator electrical connsecurity. (Access panel 2, Fig 1-4).	ections for			
1,3,5	4. Generator.				
ALL	5. Alternator.				

"FOD REMINDER"
Check work area for tools and part after completion of maintenance and inspection.

Phase		eme and CABI		A	Ircraft Serial No.	Date	3
inepect Phase No.	Inspection Requirements	Status	Faults and/or R	emarks	Action Taken		Initial
ALL C	 Perform spring scale check on collective and cyclic hydraulic actuator support mount bearings P/N 204-076-168-1 (not required on cylinder assembly P/N 205-076-099). 						
ALL	 Transmission housings, fittings, and oil manifold for chafing damage and leaks. Check vent on top of transmission case for clogging. (Access panel 2, Fig 1-4). 						
ALL C	8. Visually inspect the Kamatics Main Drive Shaft.						
6	9. Deleted.						

"FOD REMINDER"

Check work area for tools and part after completion of maintenance and inspection.

РНА	SE NO		Area Name and No.	Aircraft Serial No.	Date
1 1173	JE 110	MAIN RO	TOR AND MAST AREA 9		
Inspect Phase No.'s	Inspection Requirements	Status	Faults and/or Remarks	Action Taken	Initia
ALL C	Break torque on stabilizer bar support mount bolts (I each) and torque to specifications.				
ALL C	2. Check for excessive play in trunion bearings, collective lever bearings, and for excessive play between collective sleeve drive plate and mast.				
ALL C	Scissors and sleeve assembly for visible damage and security. Bearing and busshings for excessive play.				
ALL C	4. Disconnect scissors drive links from trunnions. Check condition of bolt shank. Check for excessive play between bolt shank and trunnion bearing uniball bore. Check swashplate bearing for roughness, binding and vertical play.				
ALL C	5. Hub spring assembly for security, condition, deformation, and cracks in rubber bumpers.				

"FOD REMINDER"
Check work area for tools and parts after completion of maintenance and inspection.

Phase	Phase No MAIN ROTOR AND I					Aircraft Serial No.	Date	
Inspect Phase No.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken		Initial
ALL	6.	Visually inspect composite main rotor blades						
C		for evidence of debonding of the leading edge abrasion strip, trim tab, and taco patch. Inspect						
		Visually inspect composite main rotor blades for evidence of debonding of the leading edge abrasion strip, trim tab, and taco patch. Inspect tie down plate and aft root weights for security. Inspect leading and trailing edge for dents and nicks.						
		dents and nicks.						
	7.	Inspect metal main rotor blades.						
C			<u> </u>					
							–	
4	8.	Remove tip cap, check stud retention nuts for looseness. Check studs for looseness or						
		looseness. Check studs for looseness or distortion.						
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"FOD REMINDER"

Check work area for tools and part after completion of maintenance and inspection.

PH	ASE	E NO	Area N ENGINE AIR INDUCT	lame an		Aircraft Serial No.		Date	ę
Inspect Phase No's		Inspection Requirements			Faults and/or F	r Remarks Action Tak			Initial
ALL C	1.	 Air induction baffle assembly for chafing, cracks, dents loose or missing fasteners and security. (Access Panel 4, Fig 1-4.) 							
ALL C	2.	 Particle separator disassembled and in- spected for clogging and damage. Gaskets and seals for cuts deterioration and separa- tion from backing plates. (Access Panel 4, Fig 1-4). 							
ALL C	3.	separator) remove	self-purging particle d, cleaned and inspected ss Panel 4, Fig 1)4.)						
3,6	4.	4. Electrical wiring for chafing deterioration of insulation and connector seals, and security of connections. (Access Panel 4, Fig 1-4.)							

PH	ASE NO.		Area Name and No. ENGINE AIR INDUCTION AREA - 10 (CONT)		Aircraft Serial No.		Date	2	
Inspect Phase No a		Inspection F	Requirements	Status	Faults and/or Remarks		Action Taken		Initial
l	ļ	CAU	TION						
l	ł	(T) e .1							
ŀ			next inspection re- igine variable inlet						
	l		be positioned to the						
		full open positi	ion. To preclude						
1		damage/distortion to VIGV components, release torques on "B" nuts to CYL 1 and CYL 2 lines at actuator.							
	l								 -
									
									
ALL C	5.		inlet housing, variable inlet es and first stage compressor foreign object damage, erosion, h deposits and oil streaks. (Ac-			<u> </u>			
									 -
]]									
		cess Panel 4, Fig 1-			·				
<u> </u>								9	
		//	.:L1. d						
ALL C	6.	Top air filter for visible damage, cleanness, condition and security of seals around edges. (Access Panel 4, Fig 1-4.)	 		·				
									
j 1		J , , , , , , , , , , , , , , , , , , ,	. . .						
									
									
ALL	7.	Right air filter fo	r visible damage clean-					·	
C	•••		and security of seals						
Ĭ			ess Panel 4, Fig 1-4.)						
									
LI	L								

Phase No.		Area N ENGINE AIR INDUC	lame an		Α	ircraft Serial No.	Date
Inspect Phase No.	Inspection Requirements		Status	Faults and/or R	emarks	narks Action Taken	
ALL C	8. Left air filter for visible damage, cleanness, condition and security of seals around edges (Access Panel 4, Fig. 1-4).						
ALL C	9. IMPROVED PARTICLE SEPARATOR						
	`						

"FOD REMINDER"
Check work area for tools and parts after completion of maintenance and inspection.

Phase No ENGINE COMPARTM		lame and No. MENT — 11		Alı	craft Serial No.	Date		
Inspect Phase No.		Inspection Re	quirements	Status	Faults and/or	Remarks	Action Taken	Initial
ALL C		1. Engine airbleed actuator strainer for condition and cleanliness. Bleed band assembly for bends, cracks and security. (Access panels 6 and 7, Fig 1-4).						
1,3,5	2.	2. Starter-generator.						
ALL C	3.	Fuel control inlet stra (Access panel 7, Fig 1-	iner inspect and clean. 4).					
ALL C	4.	Inspect and clean fuel c replace filter. (Access p	ontrol servo strainer and panel 7, fig 1-4).					
	5.	Deleted						

"FOD REMINDER" Check work area for tools and part after completion of maintenance and inspection.

PHASE NO ENGINE COMPARTME				Aircraft Serial No.	Date		
		(nepection (Negativa enta	Status	Fands and/or Remarks	Action Tobaca	termer
ALL C	6.	leaks and securi	and hoses for chafing, ity. Braided hoses for wires. (Access Panesl 6				
ALL C	7.	7. Main fuel filter micronic paper element in- spected and replaced. Clean if metal screen type. (Access Panel 7, Fig 1-4.)					
ALL C	8.	valve (1 each) outle (crashworthy only) and cracks in brea	and breakaway type of main fuel strainer for leakage, security akable (necked) section way pins in outer sleeve of Panel 7, Fig 1-4.)				
ALL C	9.	main fuel strainer security, leakage as (necked) section as	elve (1 each) on inlet of (crashworthy only) for ad cracks in breakable and play in breakaway to staked area. (Access				
ALL C	10.	oil tank inlet from (crashworthy only) and cracks in brea	alve (1 each) on engine engine breather hose for security, leakage akable (necked) section way pins in outer sleeve				

Phase	No.	77107777 00707170	Name and			Aircraft Serial No.		te
Inepect Phase No.		Inspection Requirements	Status	Faults and/or f	Remarks	Action Taken		Initial
ALL	10.A	. Self-sealing oil system component hoses for						
С	appe	activation (e.g., swelling, blistering, areas that ar to be soaked with oil).						
	!							
				·				
ALL C	11.	Engine Oil Filter Inspect (TM 55-2840-229-23).						
		<i>-,</i>						
ALL	12.	Preskaway time value (Leach) at union of						
C	12. Breakaway type valve (1 each) at union of scavenge pump outlet hose and thermal bypass input line (crashworthy only) at engine deck							
	for security, leakage, and cracks in breakable	for security, leakage, and cracks in breakable (necked) section and play in breakaway pins in						
	1-4).	outer sleeve area. (Access panels 7 and 53, Fig.						
	17.							
ALL C	13.	Breakaway type valve (1 each) on engine oil tank outlet to engine (crashworthy only) for						
	 	leakage, security, and cracks in breakable (necked) section and play in breakaway pins in]					
		outer sleeve staked area.						ļ
246	1.4							
2,4,6	14.	Engine mount rod ends for maximum altowable axial and radial play (Access Panel						
	•	7, Fig 1-4).						

"FOD REMINDER"

Check work area for tools and part after completion of maintenance and inspection.

PHASE	NO	ENGINE C	Area Name and No. COMPARTMENT -11 (Cont)	Aircraft Serial No.	Date
Inspect Phase No's	Inspection Requirements	Status	Faults and/or Remarks	Action Taken	Initial
2,4,6,	15. Engine mount pillow block assemblies for wear and damage. Trunnion caps for damage and security. Trunnion bearings for wear and excessive axial and radial play (Access Panels 6 and 7, Fig. 1-4).				
2,4,6,	16. Engine work platform decks for bonding separation, cracks, punctures and corrosion (Access Panels 7 and 10, 19 and 22, Fig. 1-4).				
ALL	17. Inspect mono, biped, and tripod engine deck mounting pads and attaching hardware for; looseness and security. If looseness is evident, check bolts and holes for damage.				
ALL	18. Engine deck drain holes and channels for obstruction.				
ALL	18.1 Inspect Break-Away Coupling (90-degree and straight halves) at the ODDS Lubriclone Filter. Check wear on the Break-Away Pins.				

"FOD REMINDER"

Phase	e No	Area N ENGINE COMPART	lame an	d No. ' — 11 (CONT)		Aircraft Serial No.	Dat	te
Inspect Phase No.	Inspection F	Requirements	Status	Faults and/or i	Remarks	Action Taken		initial
ALL	19. Power turbine gover	rnor control tube, levers, points for wear, security,						
	and corrosion.	points for wear, security,						
	ĺ							
								<u> </u>
								
ALL	20. Droop compensator lube, and corrosion.	for proper attachment,						
ALL	21. Linear actuator for security, electrical connections, wear and proper operation.							
Ì	connections, wear an	d proper operation.						
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"FOD REMINDER"
Check work area for tools and part after completion of maintenance and inspection.

Phase	No	Area CABIN ROOF — 12	Name an	d No.		Aircraft Serial No.	Dat	ie
Inspect Phase No.	Inspection R	lequirements	Status	Faults and/or I	Remarks	Action Taken		Initial
	NO	OTE						
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"FOD REMINDER"
Check work area for tools and part after completion of maintenance and inspection.

Phase	No	Area N CABIN EXT SIDES, GEAR – 13	ame and BOTT			Aircraft Serial No.	Date	e
Inspect Phase No	Inspection R	equirements	status	Faults and/or F	Remarks	Action Taken		Initial
ALL	hoses for activation (e.gareas that appear to be	em component lines and g. swelling", blistering and e soaked with fuel or have es). (Access panel 50. Fig						
ALL	bottom of aft fuel cell in breakable (necke breakaway pins in ou only to helicopter eq	res (4 each) located at s, for chafing and cracks and section and play in the staked area. (Applies uipped with crashworthy panels 48, 50, 52, 53, 55						
ALL	3. Breakaway type valves on aft end of right and left hand forward fuel cells (crashworthy only) for leakage and cracks in breakable (necked) section and for play in breakaway pins in outer staked area. (Access panels 48, 50, 52, 53, 55 and 57, Fig 1-4).							
ALL	4. Fuel lines and fittings of forward end of left and right hand forward fuel cells for leaks and chafing. (Access panels 85 and 86, Fig 1-4).							

"FOD REMINDER"

Check work area for tools and parts after completion of maintenance and Inspection.

PHASE	NO	CABIN E GEAR - 1	Area Name and No. XT SIDES, BOTTOM, & LANDING 13 (CONT)	Aircraft Serial No.	Da	te
Inspect Phase No's	Inspection Requirements	Status	Faults and/or Remarks	Action Taken		Initial
3,6	5. Landing gear crosstubes for exxcessive spread. (Check measurements).					
ALL C	6. Fuel boost pumps (electrical and air drive) visually for leaks, damage and security. (access panel 81, Fig 1-4)					
ALL C	7. Retention cap assemblies and bearing Support fittings at crosstube to fuselage, attach points for deterioration, wear an d security.					
3,6	8. External stores support assembly beams when installed, for fatigue cracks within 6 inches of upper fuselage attach points. Use non-destructive (TM 55-1500-335-23 method).					
ALL C	Landing gear skid shoes for wear, damage and security.					
ALL C	10. Landing gear skid tubes (skid shoes removed), saddles, steps, tow rings, end caps and ground handling wheel attach lugs for cracks, damage and security.					

2-36

PHASE	NO	UNDE	Area Name and No. R CABIN PYLON (HELL HOLE)-14	Aircraft Serial No.	Da	ate
Inspect Phase No's	Inspection Requirements	Status	Faults and/or Remarks	Action Taken		Initial
ALL C	1. Fuel system components and associated lines and hoses for chaffing, damage, leaks and security. (Access panels 48, 50, 52, 57 and 59, Fig. 1-4.)					
ALL C	2. Self-sealing fuel system component lines and hoses for activation (e.g., swelling, blistering, areas that appear to be soaked with fuel or have fuel between fabric plies). (Hell Hole)					
ALL C	3. Fuel supply and particle separator discharge lines and highest protruding attaching hardware in the area of transmission sump and tail rotor shaft for 0.5 inch clearance with sump. (Hell Hole)					
ALL C	4. Transmission oil lines and hoses for chafing, damage and leaks. Transmission oil line quick disconnects for excessive play.					
ALL C	5. Transmission lower housing and fittings for chafing, damage and leaks. (Hell Hole)					

"FOD REMINDER"

Phase	No.		Area N UNDER CABIN PYL (CONT)	lame and N ON (HEI	= -		Aircraft Serial No.	Date	e
inepect Phase No.		Inspection R	equirements	Status	Faults and/or R	emarks	Action Taken		Initial
	6.	Deleted							
•]					· · · · · · · · · · · · · · · · · · ·			
ALL	7.	Transmission extern	nal oil filter element						
С	ŀ	replaced, (Hell He ODDS).	ole). (Aircraft Without		·				
ALL	7.1	Inspect transmission debris monitor, aircraft with ODDS.		-					ļ . ————
C	' ''	aircraft with ODDS	nspect transmission debris monitor, aircraft with ODDS.						
•									
ALL C	8.	Bearings, bushings control linkages security. (Hell Hole)	and rod ends in flight for excessive play and						
		security. (Hell Hole)							
	<u> </u>	· · · · · · · · · · · · · · · · · · ·	····						
ALL C	9.	Flight control links	ages including push-pull		, , , , , , , , , , , , , , , , , , , ,				
		assemblies, etc., for security. (Hell Hole)	ks, idlers, support corrosion, damage and						<u> </u>
		security. (Hell Hole).							
246	10	10. Throttle control linkage for damage, wear and security. (Hell Hole).							
2.4.6 C	10.								

"FOD REMINDER"
Check work area for tools and part after completion of maintenance and inspection.

i		NO	Area N UNDER CABIN PYLON	ame an		A	ircraft Serial No.	Date	P
Inspect Phase No a		Inspection I	Requirements	Status	Faults and/or	Remarks	Action Taken		Initial
2,4,6	11.	Electrical wiring f	or chafing deterioration						
		security of connect	l connector seals, and ions. (Hell Hole)	}					
									
				,					
2,4,6	12.	Cargo suspension	assembly for damage						
С		and security. (Hell	Hole)						
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PH	ASE	NO	Ares N MID-FUSELAGE UND	ame and N ER ENG D		, A	Aircraft Serial No.	Date	e
Inspect Phase No's		Inspection	Requirements	Status	Faults and/or	Remarks	Action Taken		Initial
ALL	1.	Fuselage structurengine deck for design sion. (Access pane	e behind cabin and below amage, cracks and corro- el 53, Fig. 1-4).						
ALL C	2.	Throttle control I and security. (Acc	inkage for damage, wear ess panel 53, Fig.1-4).						
ALL	3.	Electrical wiring for chafing, deterioration of insulation and connector seals, and security of connections. (Access panel 53, Fig.1-4).							
2,4,6 C	4.	Engine idle solenoid for operation, freedom of plunger, corrosion and security. Check for proper clearance.							
ALL C	5.	Bleed air lines for (Access panel 53, F	Bleed air lines for condition and security Access panel 53, Fig. 1-4).						

L.		NO	Area N MID-FUSELAGE UNDI	ame ar	1 M 53-1320-210-P M nd No. G DECK - 15 (CONT)	A	ircraft Serial No.	Date	e
Inspect Phase No a		Inspection (Requirements	Status	Faults and/or F	Remarks	Action Taken		Inilial
		Deleted.							
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						· · · · · · · · · · · · · · · · · · ·			
ALL	7.	Fuel system lines	and hoses for chafing,						
С	<u> </u>	frayed or broken	ty. Braided hoses for wires. (Access panel 53,						ļ
		Fig.1-4).	•						
									
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	ASE NO	Area N ELECTRONIC/COMM			A	ircraft Serial No.	Date	?
Inspect Phase No's	Inspection	Requirements	Status	Faults and/or	Remarks	Action Taken		Initial
3,6	1. Fuselage structure corrosion. (Acces 1-4).	re for damage, cracks and s panels 19, 21 and 22, Fig.						
	Deleted.							
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		E NO	Area N ENGINE AREA EXTEI	lame ar RIOR -		A	ircraft Serial No.	Date	2
Inspect Phase No's		Inspection I	Requirements	Status	Faults and/or	Remarks	Action Taken		leilini
ALL	1.	Engine exhaust ta	ilpipe for cracks, dents,						
•		and burned or buck 8, Fig.1-4).	kled areas. (Access panel	 					
		0,1 1g.1 -4/.		 					
				ļ					
				-					 -
ALL	2	Uset exchange for							
ALL	٤.	and inspect for cra	r muff heater, remove acks. (Access panels 8						
		and 10, Figure 1-4).	icks. (Access panels o						
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Indian I			NO	Area N TAILBOOM INTERIOR	lame ar t - 18	nd No.	A	ircraft Serial No.	Date	e
C for corrosion, cracks and damage. (Access panels 16; 17, 18 and 36, Fig.1-4). ALL C Synchronized elevator supports (4 each) for corrosion and damage. (Access panel 17, Fig. 1-4). ALL C Synchronized elevator control linkage for damage, binding, corrosion, and loose, missing or improperly installed hardware.	Inspect Phase No's		Inspection I	Requirements	Status	Faults and/or F	lemarks	Action Taken	<u></u>	Initial
ALL C Synchronized elevator supports (4 each) for corrosion and damage. (Access panel 17, Fig. 1-4). ALL C Synchronized elevator supports (4 each) for corrosion and damage. (Access panel 17, Fig. 1-4).	ALL	1.	Tailboom structu	re, including longerons						
ALL C Synchronized elevator supports (4 each) for corrosion and damage. (Access panel 17, Fig. 1-4). ALL C Synchronized elevator control linkage for damage, binding, corrosion, and loose, missing or improperly installed hardware.	С	[for corrosion, crac	ks and damage. (Access						<u> </u>
C for corrosion and damage. (Access panel 17, Fig. 1-4). ALL C damage, binding, corrosion, and loose, missing or improperly installed hardware.	ł		panels 16, 17, 18 ar	nd 36, Fig. 1-4).	 -					
C for corrosion and damage. (Access panel 17, Fig. 1-4). ALL C damage, binding, corrosion, and loose, missing or improperly installed hardware.	1	l								
C for corrosion and damage. (Access panel 17, Fig. 1-4). ALL C damage, binding, corrosion, and loose, missing or improperly installed hardware.	1				 					ļ
C for corrosion and damage. (Access panel 17, Fig. 1-4). ALL C damage, binding, corrosion, and loose, missing or improperly installed hardware.	ΔΙΙ	2	Synchronized elev	ator supports (4 ooch)	 		,,,,,,,, 			
ALL Control of the state of the	•	 	for corrosion and d	lamage. (Access panel					·	
C damage, binding, corrosion, and loose, missing or improperly installed hardware.	1		17, Fig. 1-4).	0 (
C damage, binding, corrosion, and loose, missing or improperly installed hardware.	}									
C damage, binding, corrosion, and loose, missing or improperly installed hardware.	1									
C damage, binding, corrosion, and loose, missing or improperly installed hardware.										
missing or improperly installed hardware.		3.	Synchronized eleve	ator control linkage for	L					<u> </u>
(Access panels 17, 18 and 36, Fig.1-4).	C	ł	damage, binding,	corrosion, and loose,						
			(Access panels 17. 1	erry instance nareware. 18 and 36. Fig. 1-4).	ļ					
										
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РН	ASE	NO	1	Area Name and No. Aircraft Serial No. Date BOOM INTERIOR - 18 (CONT)		e			
Inspect Phase No's		Inspection (Requirements	Status	Faults and/or R	emarks	Action Taken		Initial
ALL C	4.	 Synchronized elevator horn assembly for corrosion, damage and proper drag with elevators removed. 							
ALL C	4.1.	 Synchronized elevators for cracks in closing (inboard) rib with elevators removed. 							
ALL C	5.	Bearings, bushings and rod ends in flight control linkages for excessive play and security. (Access panels 17, 18 and 36, Fig. 1-4).							

РН	ASE	NO	Area N TAILBOOM INTERIOR	ame and No. R - 18 (CONT)		A	ircraft Serial No.	Date	2
Inspect Phase No's		Inspection I	Requirements	Status	Fauils and/o	Remarks	Action Taken		IrelMat
ALL C	6.	tubes, links, bellc	ages including push-pull ranks, idlers, quadrant, damage and security. 18 and 36, Fig. 1-4).						
ALL C	7.	Tail rotor control broken wires and 13, 14, 16, 17 and 18	ol cables for chafing, security. (Access panels 8, Fig.1-4).						
ALL C	8.	Tail rotor control sion. (Access panel	cables for specified ten- 17, Fig.1-4).						
ALL C	9.	Control cable pulle (Access panel 16, F	ys for wear and damage. ig.1-4).						
ALL	10.	and connector seal	or chafing, deterioration ls, and security of con- anels 14 thru 18, 36 and						

"FOD REMINDER"

Phase	e No TAI	Area Nan L ROTOR DRIVE				Aircraft Serial No.	Dat	8
Inspect Phase No.	Inspection Requirer	ments S	Status	Faults and/or R	emarks	Action Taken	į	Initial
ALL	1. Intermediate (42°) gearbox	oil drained, sight						
С	gage for damage or stained (Access panel 13, Fig 1-4).	glass, and refilled.						
				····				
							<u></u>	
ALL C	2. Intermediate (42°) gearbox clogged condition. Gearbox	breather vent for						
C	for cracks, condition and panel 13, Fig 1-4).	security. (Access						
	panel 13, Fig 1-4).	<u> </u>						
		<u> </u>						
ALL C	3. Tail rotor control aft cables wires and security. (Access	for chafing, broken panels 13 and 14,		······································				
	Fig 1-4).	-						
		 						
		-		: 				
ALL	4. Control cable pulleys for	wear and damage						
C	(Access panels 13 and 14, F	Fig 1-4).						
ALL	5. Tail rotor driveshaft hange	er bearings. Inspect						
С	IAW TM 55-1520-210-23.	(Inspection may be tion was performed						
	within 25 hours of current p	ohase inspection).						

"FOD REMINDER"

Check work area for tools and part after completion of maintenance and inspection.

Phase	No	Area N T/R AND GEARBOX	lame and			Aircraft Serial No.	Dat	е
Inspect Phase No.	nspect Phase Inspection Requirements Status		Faults and/or F	Remarks	Action Taken		Initial	
ALL	1. Vertical fin rib (P/N	1 204-030-827-27) along						
С	rivet row at fin station thru topmost lightning Fig. 1-4).	10.08 for cracks (access holes). (Access panel 14,						
	11g. 1-4).					<u> </u>		
			}					
ALL	Tail rotor (90°) gearbored for damage or stained	ox oil drained sight gage						
С	for damage or stained	glass, and refilled.						
ALL	3. Tail rotor (90°) gearb	oox filler cap for clogged	ļ					
С	vent.							
ALL	4. Deleted.					1		
С								
ALL	5. Tail rotor control roll damage and security.	ler chain and sprocket for						
С	damage and security. Fig. 1-4).	(Access panels 14 and 29,						
	_							

"FOD REMINDER" Check work area for tools and part after completion of maintenance and inspection.

Phase	e No 117	Area N R AND GEARBO	lame and No. X - 20 (CO	NT)	Aircr	aft Serial No.	Da	ate
Inspect Phase No.	Inspection Requi	rements	Status	Faults and/or R	emarks	Action Taken		Initial
ALL C	6. Slowly operate tail rot and observe roller chain ensure no binding or clir sprocket occurs (Access)	operation to nbing on the				<u></u>		
ALL C	7. Remove tail rotor cont for excessive grease on t threads for wear. Thread particles (Access Panel 2	ube. Splines and ls for brass metal						
ALL	8. Tail rotor control quill corrosion, leakage and so Panel 29, Fig. 1-4).							
ALL C	9. Tail rotor control quill splines which engage qu on thread which engages (pitch control assembly r gearbox) (Access Panel 2	ill housing and s control nut removed from						
3,6 C	10. Tail rotor (90°) gearb (casting) for security and chafing by vertical fin do	l evidence of						

"FOD REMINDER"

Check work area for tools and parts after completion of maintenance and inspection.

Phase	No	Area N T/R AND GEARBO	lame and X - 20		A	ircraft Serial No.	Date
Inspect Phase No.	Inspection I		Status	Faults and/or R	emarks	Action Taken	Initial
ALL	11. Tail rotor asseml	bly balanced.					
		İ					

"FOD REMINDER"
Check work area for tools and parts after completion of maintenance and inspection.

PH.	ASE	: NO	Area N OIL COOLER/AFT BA			A	rcraft Serial No.	Date	
Inspect Phase No's		Inspection F	Requirements	Status	Faults and/or F	Remarks	Action Taken		Initial
3,6	1. Remove oil cooler screens and clean fan blades and assembly. Check fan blades for cracks. (Access panel 36, Fig. 1-4).								
ALL C	2.	 Check oil coolers for obstructions. Check turbine for rough or binding bearings by turning turbine by hand. 							
ALL C	3.	pull tubes, links, bushings and rod e	Flight control linkages, including push- pull tubes, links, bellcranks, bearings, bushings and rod ends for excessive play, corrosion, damage and security.						
ALL	4.	 Electrical wiring for chafing, deterioration of insulation and connections. 							
6	5.	Remove structural tube. Check tube and end fittings for loose rivets, cracks, corrosion and elongation of holes.							

L	HEATER COMPARTS		lame and		A	ircraft Serial No.	Date	е	
Inspect Phase No's		Inspection F	lequirements	Status	Faults and/o	r Remarks	Action Taken		Initial
3,6	corrosion. (Access panels 34 a		for damage, cracks and panels 34 and 35, Fig.1-				·		
1		4).							
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6	5.	Combustion heater	gnitor plug for condi-						-
		tion and security. (A. Fig. 1-4).	Access panels 34 and 35,						
		r ig.1-4).							
									

"FOD REMINDER"

	PHASE NO. HEATER COMPARTS					rcraft Serial No.	Dale	•
Inspect Phase No's	Inspection (Requirements	Status	Faults and/or	Remarks	Action Taken		Initial
6	6. Combustion heat assembly for cond cess panels 34 and	er radiator and jacket dition and security. (Ac- 35, Fig.1-4)						
3,6	7. Muff heater overheat switch for loose connector pins, corrosion and damage.							
	·							

1	LUBRIC	Area Name ar	nd No.	A	rcraft Serial No.	Dale
Inspect Phase No s	Inspection Requirements	Status	Faults and/or R	emarks	Action Taken	Initial
ALL C	 Lubricate in accordance wit chart contained in the manual. 					

PHASE	NO	POWER	Area Name and No. ON CHECKS	Aircraft Serial No.	Da	ate
Inspect Phase No's	Inspection Requirements	Status	Faults and/or Remarks	Action Taken		Initial
ALL C	Cyclic and collective cylinders and connecting hydraulic lines for leaks.					
ALL C	Fuel lines for leaks during engine operation.					
ALL C	Tail rotor balanced, if not previously accomplished in Area 20.					
3,6 C	Perform functional test on bleed air heater/muff heater system.					
ALL C	5. Functional test windshield wiper motor and converter assembly.					

Phase	e No POWER (Area Name and ON CHECKS (CO		,	Aircraft Serial No.	Date	
inspect Phase No.	Inspection Requirements	Status	Faults and/or F	Remarks	Action Taken		Initial
i	I .	tity indicator					
3, 6 C	6. Perform inspection on fuel quant and 20 minute fuel caution light readings.	t for correct					

				***			<u> </u>
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"FOD REMINDER"

Check work area for tools and part after completion of maintenance and inspection.

Phase	No	Area N FINAL INSPECTION	lame and			Aircraft Serial No.	Dat	e
igepect Phase No.	Inspection R	equirements	Status	Faults and/or F	temarks	Action Taken		Initial
ALL C	1. Verify that all ent accordance with Tab Forms 2408-13/14 as a	ries are completed in the 1-2 and initiate DA						
		FF						
ALL C								
				;	· · · · · · · · · · · · · · · · · · ·			
ALL C	2.1 Perform a daily inspe TM 55-1520-210-PM	ection in accordance with						
		·						
	•							
ALL C	3. Release aircraft from accomplishment of po	inspection status to permit st inspection maintenance ecordance with 5-1520-242-MTF and						
	requirements of TM 5 TM 1-1500-328-23.	5-1520-242-MTF and						
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"FOD REMINDER"

Check work area for tools and part after completion of maintenance and inspection.

By Order of the Secretary of the Army:

E. C. MEYER General, United State Army Chief of Staff

Official:

ROBERT M. JOYCE Major General, United States Army The Adjutant General

DISTRIBUTION:

To be distributed in accordance with DA Form 12-31, PM requirements for UH-1D/H and EH-1H aircraft.

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These are the instructions for sending an electronic 2028

The following format must be used if submitting an electronic 2028. The subject line must be exactly the same and all fields must be included; however only the following fields are mandatory: 1, 3, 4, 5, 6, 7, 8, 9, 10, 13, 15, 16, 17, and 27.

From: "Whomever" < whomever@ wherever.army.mil>

To: 2028@redstone.army.mil

Subject: DA Form 2028

1. *From:* Joe Smith

2. Unit: home

Address: 4300 Park
 City: Hometown

5. *St:* MO6. *Zip:* 77777

7. Date Sent: 19-OCT-93
 8. Pub no: 55-2840-229-23

9. **Pub Title:** TM

10. Publication Date: 04-JUL-85

11. Change Number: 7
12. Submitter Rank: MSG
13. Submitter FName: Joe
14. Submitter MName: T
15. Submitter LName: Smith

16. **Submitter Phone:** 123–123–1234

17. **Problem: 1** 18. *Page:* 2

19. Paragraph: 3

20. *Line:* 4 21. *NSN:* 5

22. Reference: 6

23. *Figure:* 7

24. *Table:* 8

25. Item: 9

26. *Total:* 123

27. **Text:**

This is the text for the problem below line 27.

RECOMMENDED CHANGES TO PUBLICATIONS AND BLANK FORMS

For use of this form, see AR 25-30; the proponent agency is ODISC4.

Use Part II (reverse) for Repair Parts and Special Tool Lists (RPSTL) and Supply Catalogs/ Supply Manuals (SC/SM)

DATE

8/30/02

TO: (Forward to proponent of publication or form)(Include ZIP Code)

Commander, U.S. Army Aviation and Missile Command

ATTN: AMSAM-MMC-MA-NP Redstone Arsenal, AL 35898

FROM: (Activity and location)(Include ZIP Code)

MSG, Jane Q. Doe 1234 Any Street

Nowhere Town, AL 34565

PART 1 - ALL PUBLICATIONS (EXCEPT RPSTL AND SC/SM) AND BLANK FORMS

						EXCEPT RPSTL AND SC/SM) AND BLANK FORMS		
		RM NUMBER 5–433–2				16 Sep 2002	TITLE Organizational, Direct Support, And General Support Maintenance Manual for Machine Gun, .50 Caliber M3P and M3P Machine Gun Electrical Test Set Used On Avenger Air Defense Weapon System	
ITEM	PAGE	PARA-	LINE	FIGURE	TABLE		3	
NO.	NO.	GRAPH	NO. *	NO.	NO.	RECOMMENDED CHANGES AND REASON		
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						•		
						Y	•	
				•				
			4					
				Y				

* Reference to line numbers within the paragraph or subparagraph.

MSG, Jane Q. Doe, SFC

TYPED NAME, GRADE OR TITLE

TELEPHONE EXCHANGE/ AUTOVON, PLUS EXTEN-SION

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TO: (Forward direct to addressee listed in publication) Commander, U.S. Army Aviation and Missile Command ATTN: AMSAM-MMC-MA-NP Redstone Arsenal, AL 35898						FROM: (Activity and location) (Include ZIP Code) MSG, Jane Q. Doe 1234 Any Street Nowhere Town, AL 34565						8/30/02
PUBLIC	CATION N		II - REPAIR PARTS AND	SPECIA	DATE	LISTS AN	ID SUPI	TITLE	ALOGS/	SUPPLY N	IANUAL	S
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RECOMMENDED CHANGES TO PUBLICATIONS AND BLANK FORMS For use of this form, see AR 25-30; the proponent agency is ODISC4.						cial Tool	Use Part II (reverse) for Repair Parts and Special Tool Lists (RPSTL) and Supply Catalogs/ Supply Manuals (SC/SM)					
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PUBLICATION/FORM NUMBER						DATE		TITLE	,			
ITEM NO.	PAGE NO.					RECO	DMMENDED CHANGES AND RE	ASON				
			* R	eference to li	ne number	s within the parac	araph	or subparagraph.				
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PUBLICATION NUMBER					DATE			TITLE				
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			blank forms. Add	litional blar	nk shee	ets may be	used if I	more spac	e is nee	eded.)		
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The Metric System and Equivalents

Linear Measure

- 1 centimeter = 10 millimeters = .39 inch
- 1 decimeter = 10 centimeters = 3.94 inches
- 1 meter = 10 decimeters = 39.37 inches
- 1 dekameter = 10 meters = 32.8 feet
- 1 hectometer = 10 dekameters = 328.08 feet
- 1 kilometer = 10 hectometers = 3,280.8 feet

Weights

- 1 centigram = 10 milligrams = .15 grain
- 1 decigram = 10 centigrams = 1.54 grains
- 1 gram = 10 decigram = .035 ounce
- 1 decagram = 10 grams = .35 ounce
- 1 hectogram = 10 decagrams = 3.52 ounces
- 1 kilogram = 10 hectograms = 2.2 pounds
- 1 quintal = 100 kilograms = 220.46 pounds
- 1 metric ton = 10 quintals = 1.1 short tons

Liquid Measure

- 1 centiliter = 10 milliters = .34 fl. ounce
- 1 deciliter = 10 centiliters = 3.38 fl. ounces
- 1 liter = 10 deciliters = 33.81 fl. ounces
- 1 dekaliter = 10 liters = 2.64 gallons
- 1 hectoliter = 10 dekaliters = 26.42 gallons
- 1 kiloliter = 10 hectoliters = 264.18 gallons

Square Measure

- 1 sq. centimeter = 100 sq. millimeters = .155 sq. inch
- 1 sq. decimeter = 100 sq. centimeters = 15.5 sq. inches
- 1 sq. meter (centare) = 100 sq. decimeters = 10.76 sq. feet
- 1 sq. dekameter (are) = 100 sq. meters = 1,076.4 sq. feet
- 1 sq. hectometer (hectare) = 100 sq. dekameters = 2.47 acres
- 1 sq. kilometer = 100 sq. hectometers = .386 sq. mile

Cubic Measure

- 1 cu. centimeter = 1000 cu. millimeters = .06 cu. inch
- 1 cu. decimeter = 1000 cu. centimeters = 61.02 cu. inches
- 1 cu. meter = 1000 cu. decimeters = 35.31 cu. feet

Approximate Conversion Factors

To change	То	Multiply by	To change	То	Multiply by
inches	centimeters	2.540	ounce-inches	Newton-meters	.007062
feet	meters	.305	centimeters	inches	.394
yards	meters	.914	meters	feet	3.280
miles	kilometers	1.609	meters	yards	1.094
square inches	square centimeters	6.451	kilometers	miles	.621
square feet	square meters	.093	square centimeters	square inches	.155
square yards	square meters	.836	square meters	square feet	10.764
square miles	square kilometers	2.590	square meters	square yards	1.196
acres	square hectometers	.405	square kilometers	square miles	.386
cubic feet	cubic meters	.028	square hectometers	acres	2.471
cubic yards	cubic meters	.765	cubic meters	cubic feet	35.315
fluid ounces	milliliters	29,573	cubic meters	cubic yards	1.308
pints	liters	.473	milliliters	fluid ounces	.034
quarts	liters	.946	liters	pints	2.113
gallons	liters	3.785	liters	quarts	1.057
ounces	grams	28.349	liters	gallons	.264
pounds	kilograms	.454	grams	ounces	.035
short tons	metric tons	.907	kilograms	pounds	2.205
pound-feet	Newton-meters	1.356	metric tons	short tons	1.102
pound-inches	Newton-meters	.11296			

Temperature (Exact)

F	Fahrenheit	5/9 (after	Celsius	\mathbb{C}
	temperature	subtracting 32)	temperature	

PIN: 017672-000